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CSMP MODELLING OF BRUSHLESS DC MOTORS

by

Steven M. Thomas

September 1984

Thesis Advisor:

A. Gerba

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SECURITY	CLASSIFICATIO	N OF THIS PAGE	(When Date	Entered)

REPORT DOCUMENTATION PAGE	READ INSTRUCTIONS BEFORE COMPLETING FORM					
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AD-A152 0	143					
4. TITLE (and Subtitle)	5. TYPE OF REPORT & PERIOD COVERED Master's Thesis;					
CSMP Modelling of Brushless	September 1984					
DC Motors						
	6. PERFORMING ORG. REPORT NUMBER					
7. AUTHOR(s)	S. CONTRACT OR GRANT NUMBER(s)					
Steven M. Thomas						
, seven Fr. Monas						
9. PERFORMING ORGANIZATION NAME AND ADDRESS	10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS					
Naval Postgraduate School						
Monterey, California 93943						
11. CONTROLLING OFFICE NAME AND ADDRESS	12. REPORT DATE					
Naval Pöstgraduate School	September 1984					
Monterey, California 93943	13. NUMBER OF PAGES					
l	15. SECURITY CLASS. (of this report)					
14. MONITORING AGENCY NAME & ADDRESS(if different from Controlling Office)	UNCLASSIFIED					
	UNCLASSIFIED					
	15a. DECLASSIFICATION/ DOWNGRADING SCHEDULE					
	SCHEDULE					
16. DISTRIBUTION STATEMENT (of this Report)						
Approved for public release, distributio	n unlimited					
·						
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different fro	an Report)					
18. SUPPLEMENTARY NOTES						
19. KEY WORDS (Continue on reverse side if necessary and identify by block number)						
Brushless DC Motors						
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20. ABSTRACT (Continue on reverse side if necessary and identify by block number)						
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CSMP Modelling of Erushless DC Motors

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Steven M. Thomas Lieutenant, United States Navy B.A., University of Delaware, 1976

Submitted in partial fulfillment of the requirements for the degree of

MASTER OF SCIENCE IN ELECTRICAL ENGINEERING

from the

NAVAI POSTGRADUATE SCHOOL September 1984

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ABSTRACT

Recent improvements in rare earth magnets have made it possible to construct strong, lightweight, high horsepower DC motors. This has cccasioned a reassessment of electromechanical actuators as alternatives to comparable pneuratic and hydraulic systems for use in flight control actuators for tactical missiles. This thesis develops a low-order mathematical model for the simulation and analysis of brushless BC motor performance. The model is implemented in CSMP It is used to predict such motor performance language. curves as speed, current and power versus torque. Electronic commutation based on Hall effect sensor rositional feedback is simulated. Steady state motor behavior is studied under both constant and variable air gap flux conditions. The variable flux takes two different forms. In the first case, the flux is varied as a simple sinusoid. In the second case, the flux is varied as the sum of a sinusoid and one of its harmonics.

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I. INTRODUCTION

Direct current electric motors are gaining wider use in space applications where long life, high reliability, high torque and light weight are critical factors. years, advances in magnet technology and materials, especially the use of rare earth magnets, have made possible significant improvements in the torque to inertia ratios of permanent magnet (PM) motors. Samarium cobalt is one such rare earth magnetic material. The great potential of the rare earth magnets derive from their inherent high flux density and high coercivity properties. Higher flux densities mean greater developed motor torque while high coercivity means greater innate resistance to demagnetization which permits thinner magnet sections. The two factors in combination result in increased mechanical power and energy product with decreased physical size and weight.

Brushless dc motors belong to this class of permanent magnet motors and thus enjoy their improved torque to size characteristics. In fact, brushless dc motors are significantly smaller than either their ac counterparts or conventional brush-type dc motors on an equal horsepower basis. Compared to its conventional counterpart, a brushless do motor's structure is "inside out." In the brushless configuration, the permanent magnets are attached to the rotor and the conducting coils are placed in the stator. As its name suggests, brushes have been eliminated from this type of motor. Instead, commutation of current in the stationary armature is accomplished electronically by switching on the appropriate windings via transistors specially designed to provide the high currents needed for motor power. In crder to sequence the current to the coils, the commutation circuitry is provided rotor position feedback from sensing

devices, most commonly from Hall effect sensors. These are simple semiconductor devices that develop a polarized voltage depending upon the magnetic field passing through them. Small, highly sensitive and reliable, Hall effect sensors require little power to operate and thus are widely used.

Brushless dc motors possess all the advantages traditionally associated with conventional dc motors such as linear torque-speed characteristics, excellent response In addition, there are a number times and high efficiency. of important advantages that brushless dc motors possess over standard brush-type dc motors. An obvious advantage is that electronic commutation means the elimination of commutators, the wear and tear of brushes, and the build up of carbon dust which heretofore significantly limited the life of the standard dc motor. Since electronic current switching essentially involves no component wear, results improved reliability and enhanced lifespan. As well as being subject to wear and tear, the action of brushes sliding over commutators also generates radio frequency interference (RFI) which is a severe liability in many applications. In fact, it is often in hostile and explosive environments that the advantages of dc motors are Another benefit is in heat transfer characteristics. By placing windings in the stator the thermal path to the environment is made shorter and more direct. With the removal of heat thus enhanced, mechanical and electrical malfunctions due to heat are reduced. A further important advantage of the brushless dc motor is in the area of motor speed variability. Whereas ac motors usually operate at one speed fixed by the powerline frequency, any dc motor's speed can be changed to meet varying power requirements by simply adjusting the dc input voltage. The electronic commutation scheme of the brushless do motor lends itself nicely to just

this type of variable speed control. It is a relatively simple matter to attach to the commutation circuit board a small, inexpensive microprocessor whose logic can be programmed to run the motor at any of several speeds or in either direction. A single motor can thus be programmed, in place if desired, so that its speed, direction and torque are matched to a particular application.

There are at least as many applications for brushless dc motors as there are for brush-type motor systems as well as a host of new applications. Of particular relevance to this thesis is the use of brushless do motors in flight control actuator sytems of Navy tactical missiles, particular, the cruise missile. There are three general types of actuators that are in use in missile systems: pneumatic, hydraulic and electromechanical. The maneuvering requirements of the more advanced tactical missiles call for high control torques which heretofore could only be produced by high pressure pneumatic or hydraulic actuators. these actuator systems are more susceptible to mechanical malfunction than their electrical counterparts. improved torque to inertia ratios resulting from better rare earth magnet materials and the higher mechanical reliability due to removal of brushes, as well as other advantages previously enumerated, the brushless dc motor is in a position to compete with most pneumatic and hydraulic systems. This thesis is but one part of a detailed study of the state of the art in electromechanical actuators as applied to the cruise missile system [Ref. 1].

II. NATURE OF THE PROBLEM

A. PROGRAMMING LANGUAGE

The objective of this thesis is to create a simplified low-order mathematical model that will accurately simulate the response of a brushless dc motor. The IBM Continuous System Modeling Program (CSMP) was chosen for use since it is a user-oriented computer language specifically created for modeling dynamic physical systems and is one of the most widely used in this area [Ref. 2]. It has the flexibility of determining the response of a system that is either in a block diagram format or as a set of ordinary differential equations. While CSMP is an offshoot of IBM's Digital Simulation Language, it embodies Fortran as its source language and retains much of Fortran's capability and CSMP is an applications-oriented program intended for use with the IBM mainframe systems. utility and ease of using this program language stems from simplified program control statements that almost exactly describe the mathematical equations or physical variables of the system and from the flexibility of its program structure which contains preprogrammed function blocks that obviate the need for complicated subroutine programming. essence, CSMP was chosen because it permits the user to concentrate on the details of the physical system rather than on the time-consuming complexities of programming.

E. SYSTEM BLOCK DIAGRAM

The block diagram format was chosen as the basis for the CSMP program since it is a convenient way of analyzing the input-output relationships of a physical system and the flow of signals through it without having to refer to the system equations. The basic idea of a block diagram stems from the application of Laplace transforms to a system's integro-differential or differential equations. In effect, a differential equation is rendered into a simpler algebraic expression which, in turn, becomes a fundamental equation of a system's functional blocks. Each block indicates the relationship between its input driving signal and output ie., the transfer function of the input and response: The transfer function is defined as the ratio of the Laplace transform of the output to the transform of the input with all initial conditions assumed to be zerc. block's transfer function does not include any information about the internal structure of that part of a system, merely the transformation of a signal between two points in Furthermore, it should be borne in mind that all physical systems have certain transfer characteristics that actually nonlinear to some extent. A dc motor system can be fairly accurately described using the transfer function approach to modeling and, when nonlinear transfer characteristics exist, CSMP modeling can usually be obtained from preprogrammed ncnlinear function routines.

C. DEVELOPMENT OF MCTOR SYSTEM EQUATIONS

In order to construct the block diagram of a system, one approach is to identify the various mathematical relationships between the components of the system and to write balance equations that will identify each individual block comprising the system. The system equations for a brushless do motor are presented in chapter three where all the details of the logic control and switching of transistors for commutation of the motor are developed. What follows is a simplified input-output characteristics model of the type

that has been a standard for brush-type dc motors used in control system studies [Ref. 3]. The development of this input-output model is the first learning step in the modeling process for the brushless dc motor and has the additional usefulness of being a tool for a concurrent mechanical engineering study of load torque requirements for the motor [Ref. 1].

To write the electrical balance equation describing a basic dc motor, the classical loop method based on Kirchoff's voltage law is applied to the equivalent motor circuit (Figure 2.1).

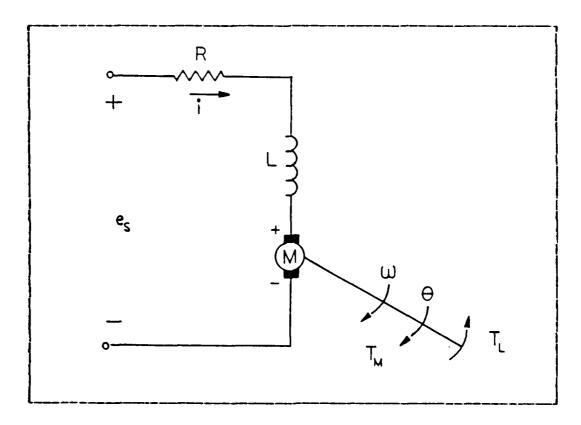


Figure 2.1 Equivalent DC Motor Circuit

The basic dc motor has the coil windings on the armature and the magnetic field scurce on the stator. In the case of the brushless dc motor, the rotor consists of permanent magnets that produce the magnetic field and the stator contains the coil windings. The Kirchoff Law for the stator circuit is

$$es(t) = Ldi/dt + Ri(t) + eb(t)$$
 (eqr. 2.1)

where R and L are the stator resistance and inductance respectively, and eb(t) is the back emf. When a conductor moves in a magnetic field, or there is any relative motion between the magnetic field source and the conductor, a voltage is generated across the terminals of the conductor. In the case of the dc motor, the voltage is proportional to the shaft velocity and tends to oppose the current flow. The relationship between the back emf and the shaft velocity is

$$es(t) = (Km*\phi)*wm(t) = Kb*wm(t)$$
 (eqn 2.2)

$$es(t) = Kb*d\Thetam(t)/dt (eqn 2.3)$$

Substituting and rearranging equations 2.1 and 2.2 into their differential forms gives the following two equations.

$$di/dt = (1/L) *es(t) - (R/L) *i(t) - (1/L) *Kb*wm(t)$$
 (eqn 2.4)

$$d\Theta m(t)/dt = wm(t)$$
 (eqn 2.5)

The basic balance equation that describes the motor as a mechanical system derives from application of Newton's laws of motion to the system components. The dynamic equation for a motor coupled to its load is

$$tm(t) = J*dwm(t)/dt + B*wm(t) + tl(t)$$
 (eqn 2.6)

where tl(t) is the load torque. Rearranging the equation it becomes

$$dwm(t)/dt = (1/J) *tm(t) - (1/J) *tl(t) - (B/J) *wm(t) (eqn 2.7)$$

B is the total viscous friction coefficient of the motor and is comprised of the sum of the viscous friction due to the motor, Bm, and the load, Bl, as seen by the motor shaft. Likewise, J is the total system inertia comprised of the sum of the motor inertia, Jm, and of the load inertia, Jl, reflected through the coupling device to the motor shaft. Both Bl and Jl are functions of the motor's speed reducing mechanism where N is the speed reduction ratio (N>1). The following equations summarize the foregoing.

$$B = Bm + Bl (eqn 2.8)$$

$$\mathbf{J} = \mathbf{Jm} + \mathbf{J1} \tag{eqn 2.9}$$

$$B1 = Blp/N (eqn 2.10)$$

$$J1 = J1p/N (eqn 2.11)$$

Jlp and Blp are the inertia and viscous friction seen at the load side of the system. Since a dc motor is essentially a torque transducer that converts electrical energy into mechanical energy, the following equation is necessary to show the relationship amoung the developed torque, tr(t), the air gap flux, ϕ , and the stator current, i(t).

$$tm(t) = (Km*\phi)*i(t)$$
 (eqn 2.12)

Because the magnetic field in the motor is assumed uniform and constant, this can be simplified to

$$tm(t) = Kt*i(t)$$
 (eqn 2.13)

where Kt is the torque constant of the motor.

Equations 2.1 through 2.13 represent the cause and effect equations of the system. The application of es(t) produces a current flow which causes the torque and the back emf to be generated. The torque produced then causes the angular displacement $\Theta m(t)$. Given the differential equations of the system, the Laplace transform can be applied and the block diagram of the system generated. As an example, the block for equation 2.1 is generated as follows

$$es(t) - eb(t) = L*di/dt + R*i(t)$$
 (eqn 2.14)

$$\mathcal{L}\{es(t) - eb(t)\} = \mathcal{L}\{L*di/dt + R*i(t)\}$$
 (eqn. 2.15)

$$En(s) = Es(s) - Eb(s) = (Ls + R)*I(s)$$
 (eqn 2.16)

$$I(s)/En(s) = Is/(Es - Eb) = 1/(Ls + R)$$
 (eqn 2.17)

Performing the same operation to the remaining differential equations, the block diagram in Figure 2.2 results.

It is important to note that while a dc motor is basically an open-loop system, the back emf of this type of dc motor acts as a natural feedback loop that tends to improve the stability of the motor.

Conce the basic block diagram is developed, writing a CSMP program is simple and fairly straightforward. Care must be taken, however, to put the block transfer functions into the proper format for CSMP's functional blocks. A copy of the CSMP program for a basic dc motor is provided in Appendix A.

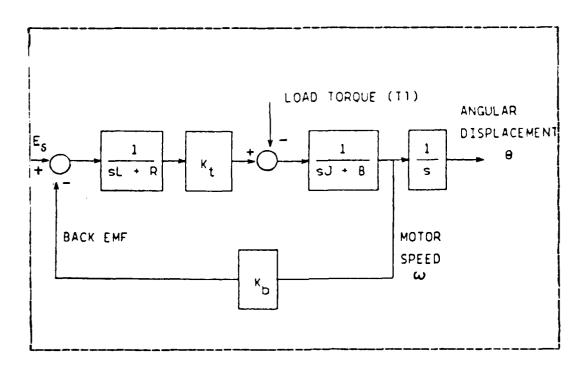


Figure 2.2 Block Diagram of DC Motor

III. CURVE PREDICTION

A. OVERVIEW

One of the purposes of this model is to produce a set of rerformance curves that accurately simulate a given motor's behavior when operated under varying load condi-As such the model can be a useful tool for studying tions. the changing performance due to the configuration changes of a motor under development or for observing the behavior of a motor under varying conditions of application or environment. Thus, having developed an input-output computer model for the DC motor, the next step was to assign values to the model parameters and run the program. Program inputs, constants and parameters were assigned values that are typical of brushless DC motors commercially available. Table I provides the parameter values for a given motor that will be used as a generic motor for the following analysis and discussion.

TABLE I
Typical Commercial Motor Parameters

Stator Resistance, R Stator Inductance, L Torque constant, Kt Back EMF constant, Kb Rotor Inertia, Jm Viscous Friction	2.74 0.0016 15.9 0.112 0.001	ohms henries oz-in/amr volt/rad/s oz-in/s²
Coefficient, Em	unknown	oz-in/rad/s

B. SPEED VS TORQUE CURVE

Specifically, three curves were produced: motor speed vs. generated torque, current vs. torque and output power vs. torque. Since permanent magnet DC motors have linear characteristics, a straight-line curve resulted under all load conditions when the parameters of Table I were inserted into the model and a voltage of 30 VDC was applied. In order to fully understand the individual contributions of the various motor parameters on the speed-torque curve, it was necessary to determine the mathematical relationships between the parameters. Referring to equations 2.1 and 2.5, the electrical and motor equations when considered under steady state conditions become

$$es(t) = P*i(t) + Kb*wm(t)$$
 (eqn 3.1)

$$tm(t) = Kt*i(t)$$
 (eqn 3.2)

Solving for i(t) and combining the equations results in

$$es(t) = (R/Kt)*tn(t) + Kb*wn(t)$$
 (eqn 3.3)

When there is no torque applied to the motor, the current is small enough to neglect and the no-load velocity becomes

$$wnl = es/Kb$$
 (eqn 3.4)

Given that the applied voltage is a constant, it is apparent that knowledge of the value of the back emf constant, Kb, will provide the various no-load speeds for any value of applied voltage. On the other hand, in the absence of any known motor specifications, the no-load speed of a given motor can be empirically measured and then the back emf constant, Kb, can be mathematically derived.

Looking next at the slope of the speed-torque curve, it can be seen that whereas one endpoint of the curve terminates at no-load speed, the other ends at stall which is that value of torque which is large enough to literally stop the motor. From equations 3.1 and 3.2, with wm set to zero the generated torque, tm(t), at stall is ts given by

$$ts = (Kt/R)*es (eqn 3.5)$$

Combining equations 3.4 and 3.5, the equation of a straight line for the speed-torque curve results in

$$wm(t) = wnl - (Rm*tm(t))$$
 (eqn 3.6)

where

$$Rm = R/(Kb*Kt)$$
 (eqn 3.7)

and Rm is the slope factor for the curve, also called the speed regulation constant. Obviously, changing any or all of the parameters will alter the slope factor, Rm. However, the back emf constant, Kb, and the torque constant, Kt, are strictly related by the common factor, air gap flux, ϕ . In fact, when both constants are put in the MKS system of units, Kb in volt/rad/s and Kt in Nm/amp, then they are equal as equation 3.8 shows.

$$Kb (volt/rad/s) = Kt (Nm/amp)$$
 (eqn 3.8)

Thus, knowledge of either constant provides knowledge of the other. Referring to equation 3.7, it can be seen that the final unknowns are the resistance, R, and the slope factor, Rm. If the resistance, R, is known from manufacturer specification sheets, for example, then the slope, Rm, can be

easily sclved. On the other hand, if given an empirically measured speed-torque curve, it is now obvious that the curve not only provides Kb which in turn provides Kt, but it also can provide a derived value for motor resistance, R. Once acquired, the three basic parameters, Kb, Kt and R, were rlugged into the CSMP model and the speed-torque curve for a 30 VDC input was generated (see Figure 3.1).

This same procedure can be used to produce a family of speed-torque curves if it is desired to study a motor's performance under different input voltages. Figure 3.2 shows the family of curves produced when the values given in Table I were entered into the model.

C. CURRENT VS TORQUE CURVE

With the speed-torque curve understood, the next task was to repeat the same general procedure for the motor current vs. torque. As a starting point, the equation for the steady-state current is given by

iss(t) =
$$(es(t) - Kt*wm(t))/R$$
 (eqn 3.9)

Since Kb and R were fixed by the speed-torque curve, generating a current-torque curve was straightforward procedure for any applied voltage (see Figure 3.3).

D. CUTPUT POWER VS TCRQUE CURVE

The final performance curve to be studied was the output power vs. torque curve. The basic equation for power is given by

$$P(t) = tm(t) *wm(t)$$
 (eqn 3.10)

$$P(t) = Kt*i(t)*wm(t)$$
 (eqn 3.11)

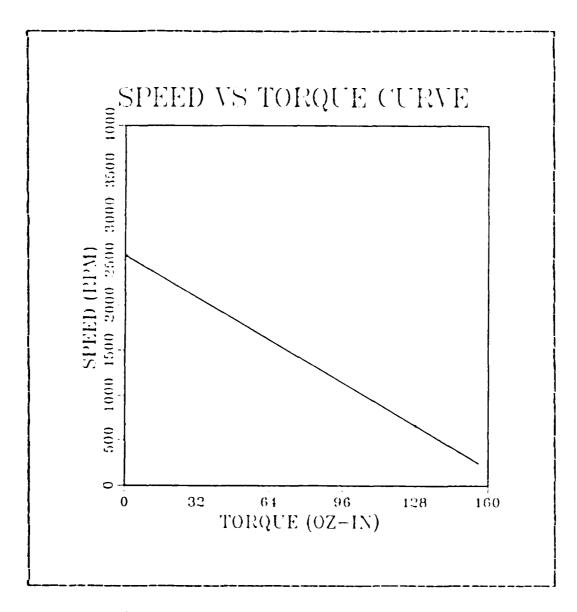


Figure 3.1 Motor Speed vs Torque Curve

where power is measured in watts. Since this equation involves no new constants, producing the power curve was only a matter of adding a line to the existing model which also included a conversion factor for changing the torque units from oz-in/amp to Nm/amp in order for the power to be

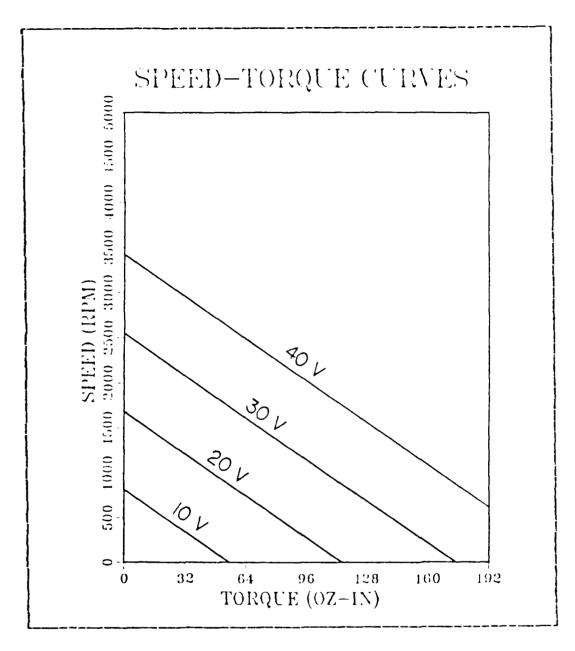


Figure 3.2 Family of Speed-Torque Curves

dimensionally correct in watts. Again running the program with the parameters given in Table I, the model produced the power curve given in Figure 3.4

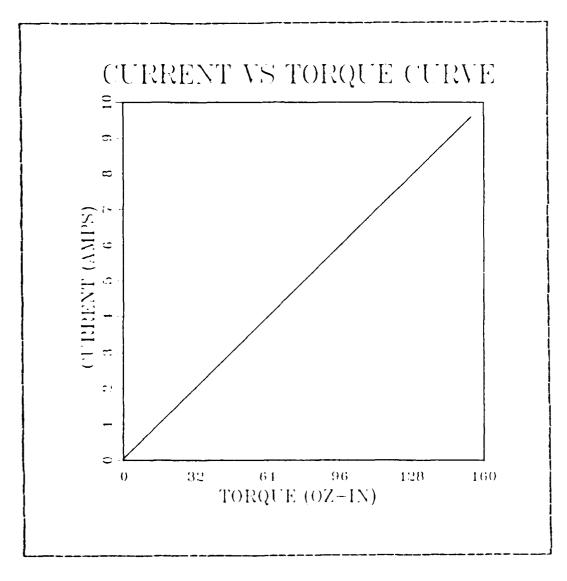


Figure 3.3 Motor Current vs Torque Curve

As expected both the speed-torque and current-torque curves were linear. The power-torque curve is obviously not linear, especially at high leads where the power begins to roll off fairly sharply. From the molelling point of view this is due solely to the fact that the motor slows down faster than the load torque increases (see equation 3.9).

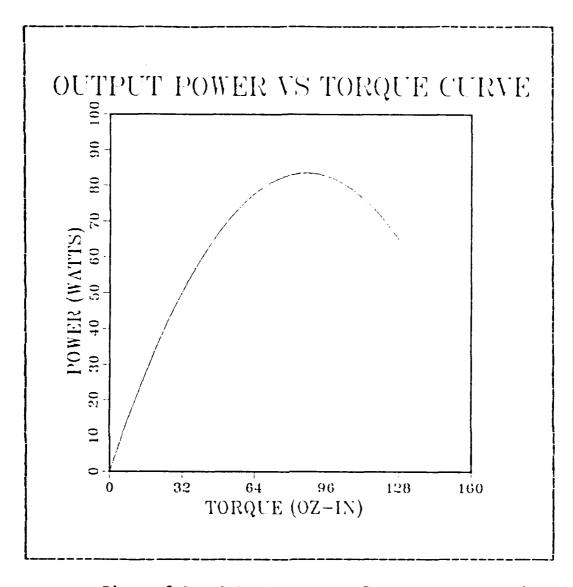


Figure 3.4 Output Power vs Torque Curve

Another factor—that further accentuates this—power rolloff in an actual motor is the effect due—to armature reaction. Whenever a current flows—through the motor armature—in a permanent magnet dc mctor,—the armature becomes an electromagnet which produces—a flux that tends to—oppose the flux produced by the permanent magnets.—This has the effect of

partially demagnetizing the permanent magnets. This demagnetization is a reversible effect, meaning that as current returns to zero the permanent magnets return to However, as the current increases and their full strength. armature reaction sets in, it has the effect of decreasing the torque constant, Kt (as well as Kb). As equation 3.9 indicates, this also contributes to the output power rolloff Though current permanent magnet IC mctors at high loads. made of rare earth magnets have high coercivity that resists armature reaction, at high levels of rated current a small amount of armature reaction still occurs. For the purposes of this thesis, the model is considered to accurately simulate motor behavior fcr loads up to the peak power load for the given applied voltage. From that load upward, assumed that armature reaction is likely to occur and result in nonlinear behavior which is not included in this model.

E. MCTOR REVERSAL

The final step in the modeling procedure was to reverse the motor's direction and ensure that mirror images of the three curves resulted. To do this, it was first necessary to replace the positive input voltage with a negative counterpart. Next, in order to maintain model consistency where the load torque, tl(t), is treated as a torque that opposes the motion of the motor, it was necessary to modify the model so that the load torque was a positive value. This consisted of creating a CSMP procedure block that identified the input voltage as either positive or negative and then treated the load torque accordingly. Once program lugs were removed, the model accurately simulated motor reversal with proper speed, current and power values for the appropriate load conditions.

IV. ELECTRONIC COMMUTATION

A. SWITCHING ACTION

The next major step in this thesis was to convert the basic model of a standard brush-type dc motor into a three-phase, four-pole brushless dc motor system. Figure 4.1 shows this system with the three-phase stator windings configured in a standard 'star' connection with each winding oriented 120 electrical degrees from the others.

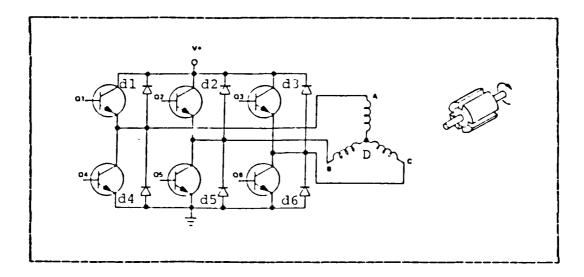


Figure 4.1 Controller Configuration

The six transistors are connected to the ends of each stator leg and through logic-controlled switching action provide three-phase full-wave motor control. Figure 4.2 indicates the switching logic sequence for counterclockwise rotation.

To produce clockwise rotation, the sequence is reversed. In either case, a pair of transistors will be switched at the start of each 30 degree (mechanical) interval which causes current to simultaneously build up in one leg, flow

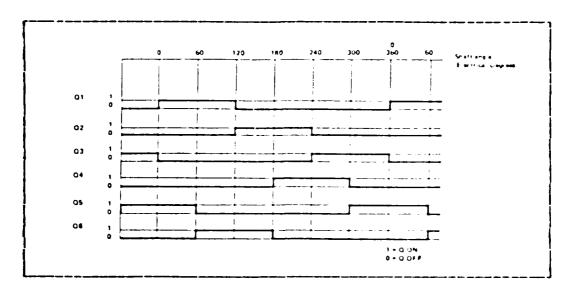


Figure 4.2 Switching Logic of a 3-Phase Brushless DC Motor

steadily in a second, and decay to zero in a third. Given proper sequencing, the net developed torque ideally reaches a steady state value that causes motor rotation at a constant speed in the desired direction.

For windings configured in a star arrangement, it can be seen that conduction is continuous in one leg while commutation occurs in the other two less. For example, as Q1 and Q5 are energized between 30 and 60 legrees (mechanical), current flows down leg A and into leg B. At the same time, the current through leg C due to the energy storage behavior of an inductor decays to zero through diode, D? (see Figure 4.1). In the next sequence, 60 to 90 degrees, Q6 is energized and Q5 switched off. Current has reached a steady state flow through leg A, but now it flows down leg 2 and the current in leg B decays to zero through D2. In the next sequence, 60 to 120 degrees, Q2 is switched on and Q1 switched off. Current has reached steady state flow through leg C but now current builds up and flows from leg B while the current in leg A decays through leg C. This same action repeats in subsequent commutation states. Looking at point C in Figure 4.1, it can be seen that, since current is always continuous in one leg whether flowing into or from node D, Kirchoff's Current Law requires that the net current flowing in the other two legs must equal the flow in the continuous leg. This permits the following important simplification which was needed later in the modeling process. Namely, the three-legged stator can be approximated by just two windings serially configured. will always accurately reflect steady state flow conditions while the other will be treated as if it were in steady state because the build up in one leg is balanced by decay in the other. It is recognized that this model is a first approximation and that future modeles will require additional refinements in order to adequately represent the effect of switching transients on the transistor and diode elements.

P. HALL EFFECT SENSOR FEEDBACK

As was mentioned in the introduction, the switching action of the commutation circuitry is based upon rotor position feedback from Hall effect sensors. In order to represent electronic switching and postional feedback, as well as the three phase winding configuration, it was necessary to modify the block diagram of Figure 2.1 to that in Figure 4.3

Comparing Figures 2.1 and 4.3, it is seen that the chief difference is in the switching logic block and the addition of two more transfer function blocks to account for the additional windings. To better understand the logic relationship between the sensors and the current switches, the truth table for both counterclockwise and clockwise rotation is presented in Table II.

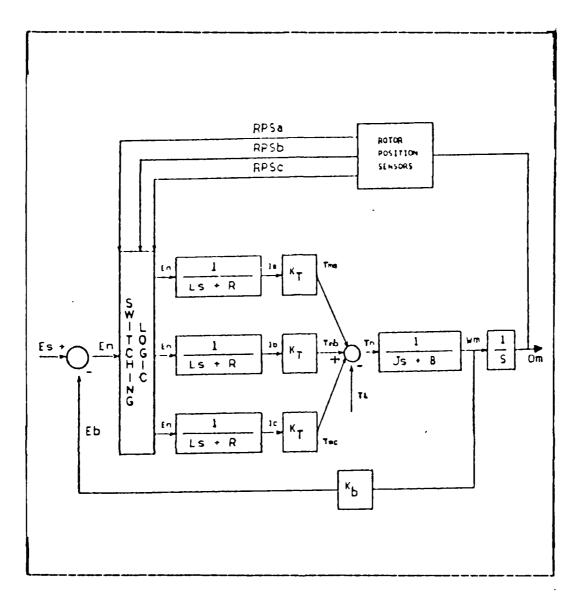


Figure 4.3 Equivalent Circuit With Electronic Commutation.

A brief review of the action af a Hall effect sensor is in order at this point. Consider a conducting material with a constant current flow. When there is no external magnetic field present, the current flows undisturbed in a straight line. But when a magnetic field is applied, a deflecting force, the Lorentz force given by Il x B (where I is the

TABLE II
Sensor and Switching Logic

RPS	A	RPS	В	RPS	C	PH	A	PH 5	PH C
10000		10001		0011		hig ope low low	h en	low open hijh open low	oren low oren high high
				COUNT	TERCLOCKW:	ISE			
0 1 1 0 0		0 0 1 1		100001		low ope hig ope low	n h h n	high high open low open	oren lown lopen high

CLOCKWISE

current, 1 is the length of the conductor and B is the magnetic field) causes the current to bend and electrons to pile up on one side and positive charges, or holes, to do the same on the other side. Thus a transverse Hall potential difference, Vh, develops across the conductor (see Figure 4.4).

Figure 4.5 shows that in the presence of a sinuscidal magnetic field produced by a rotating magnet, such as one of the pole-pairs of a multiple-pole motor, the induced Hall voltage takes on an ideal square wave shape.

The logic circuitry of the sensor is configured in such a way as to output a logic level 1 for positive Hall voltages and a 0 for negative voltages. Briefly reviewing the action of the motor: as the shaft turns, the rotor position sensors send this information to the switching logic block which then decides which phases are to be made high, low or open in accordance with Table II. Equivalently, the voltage, En, is then applied to the two appropriate windings

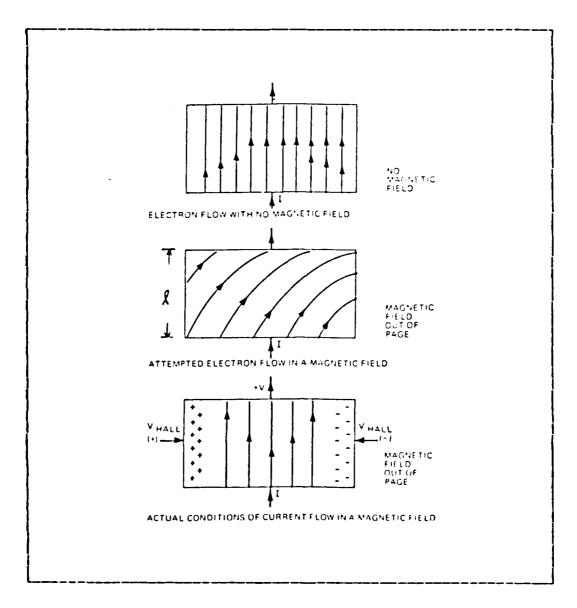


Figure 4.4 The Hall Effect

and zero voltage is applied to the third winding in agreement with the assumption that current is steady in one leg while the commutation in the other two legs is equivalent to a continuous current in one leg. As before, each voltage then produces a current. Because this is assumed to be a

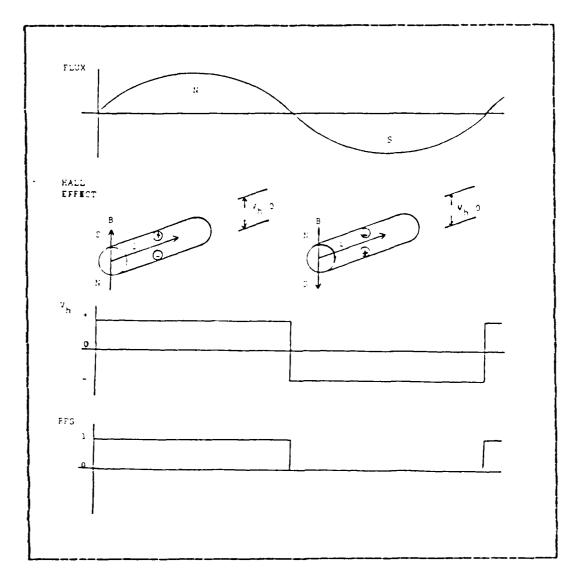


Figure 4.5 Sensor Logic Based on a Hall Effect Sensor

linear system, the principle of superposition is applicable and, thus, the torques developed by current flow through separate windings car be considered independently and then summed to give a total motor torque. Beyond this point, the brushless motor components are identical to its brush-type counterpart and behave similarly.

C. MCDEI REVISION

CSMF modeling of this version of the brushless do motor called for two modifications to the brush-type model. Essentially, a CSMF procedure function was constructed to act as a subroutine that embodies the logic of the switching action. The logic recognizes that if the motor shaft is within some degree rarge, for example 30 to 60 degrees, then it sets the appropriate rotor position sensors and energizes the switches for the appropriate two windings and deenergizes the switch for the third. In effect, only the forcing voltages are passed to the main program, whereas switch and sensor positions are available for output (see Appendix A).

V. VARIABLE FLUX

A. AIR GAP FLUX

The steady state behavior of both a brush-type and a trushless dc motor has been studied under the fundamental assumption that the lack emf generated is a product of the shaft velocity times a constant, Kb (see Equation 2.2). Likewise, the motor torque is equated to the product of the current times a constant, Kt (see Equation 2.12). Equation 2.11 reveals in the case of the torque constant, Kt, the torque is in fact the product of some other constant of proportionality, Kt1, times the air gap flux, ϕ . same is true for the back emf constant, Kb; that is, Kb is the product of some proportionality constant, Kb1, times the In both cases, the flux, ¢, has been air gap flux, 0. assumed to be constant which is in agreement with the general practice when modeling permanent magnet dc motors. In point of fact, the flux is not constant but rather is distributed sinusoidally, generally being the sum of a sirusoid and one or more of its harmonics. How the physically constructed determines this flux distribution. Two of the chief factors are the structure of the magnetic pole faces and, most importantly, how the windings are distributed in the armature. The latter includes such factors as the number of slots, the number of coils per slot and the spacing of the slots [Ref. 5]. The careful designer distributes the windings so as to minimize the effects of Regardless of their constituents, composite sinusoids have an average value, of course, and so in practice it is this average value that becomes the constant flux term, ¢.

B. FLUX AS AN AVERAGE VALUE

A logical next step in the design of a more realistic model was the determination of an average value for the varible flux, ϕ avg, and the corresponding proportionality constant, KK1 such that the ratio of ϕ avg and KK1 equals one and the relationships of equations 2.2 and 2.5 are preserved. The equations for back emf, eb(t), and developed torque, tm(t), now become

$$eb(t) = Kbp * wm(t)$$
 (eqn 5.1)

$$tm(t) = Ktp * i(t)$$
 (eqn 5.2)

where the back emf constant, Kbp, and the torque constant, Ktp, are now expressed as

$$Kbp = Kb * (\phi a vg/KK1)$$
 (eqn 5.3)

$$Ktp = Kt * (\phi a vg/KK1)$$
 (eqn 5.4)

Assuming as a first approximation that the air gap flux varies as a simple sinusoid of unit magnitude, the first step toward finding the total system flux, bavg, was to determine the relationship of the individual flux patterns in each of the three phase windings. Analysis of a fourpole magnet rotating under three windings separated by 120 degrees (electrical) revealed the following relationships

$$\phi a = \sin(2\theta + \pi/6)$$
 (eqn 5.5)

$$\phi b = \sin(2\theta + 9\pi/6)$$
 (eqn 5.6)

$$\phi c = \sin(2\theta + 5\pi/6)$$
 (eqn 5.7)

where ¢a, Øp and ¢c are the flux contributions of the respective phases. Feturning to an earlier assumption that the circuit can be approximated as two windings in series, ignoring the third, then the average flix can also be approximated from the sum of the flux developed in two windings at a time. This approximation is further supported by the fact that the flux in the decay leg has, in fact, an average value of zero during every commutation interval. From the switching sequence of Figure 4.1, the conducting legs were identified for each interval and, from Equations 5.5 through 5.7, their respective flux values calculated and summed. For every interval, the resulting flux waveshape was identical and had an average value of $\phi = 1.631$ (see Figure 5.1).

Since KK1 merely normalizes ϕ avg, its value was easily determined to be KK1 = 0.613. The appropriate substitutions were then made in the CSMP model (see Appendix A for Revision Two). Since this amounted to nothing more than a rearrangement of constants, the model's output behavior remained unchanged. At this stage, the model simulates the behavior of a brushless do motor being electronically commutated and exhibiting constant, steady state developed torque and motor speed due to a fixed, average value air gap flux.

C. SINUSOIDAL FLUX

The final stage of this thesis was to examine the steady state behavior of the brushless do motor under variable air gap flux conditions. As a first approximation, the flux was modeled as a simple sinusoid and the flux relationships are those given in equations 5.5 through 5.7 As before, only the flux relating to the two conducting legs was considered. This composite variable flux, \$v1, is shown in Figure 5.1 Equations 5.3 and 5.4 had to be modified to account for the different value of flux.

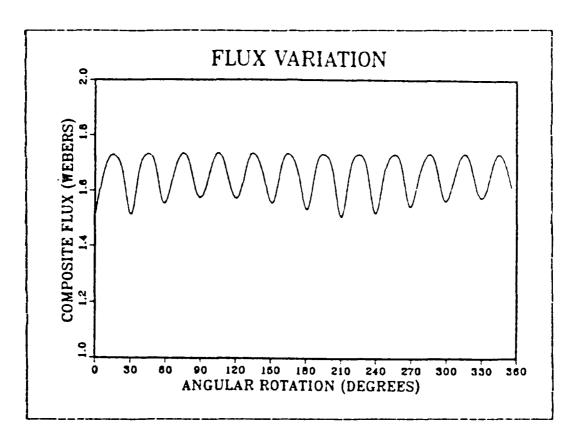


Figure 5.1 Composite Plux Variation for Two Windings

$$Kbpp = Kb * (\phi v 1/KK2)$$
 (eqn 5.3)

$$Ktpp = Kt * (\phi v 1/KK2)$$
 (eqn 5.9)

Recall that the objective was to keep the values of Kbpr and Ktpp as near as possible to those of Kb and Kt, respectively. In order to roughly normalize $\phi v1$, the constant value was determined as KK2 = 1.655. Since $\phi v1$ was now a variable quantity, it should be recognized that Kbpr and Ktpp were no longer strictly constants and were better expressed as the following functions.

$$Kbpp = Kb(\phi v 1)$$
 (eqn 5.10)

$$Ktpp = Kt (\phi v 1)$$
 (eqn 5.11)

The equations for back emf, eb(t), and developed torque, tm(t), now became

$$eb(t, \phi v1) = Kbpp * wm(t)$$
 (eqn 5.12)

$$tm(t, \phi v1) = Ktpp * i(t)$$
 (eqn 5.13)

Pecause the total flux was no longer a constant, average value, it was expected that variation in developed torque (and, consequently, in motor speed) would result. Furthermore, as the system block diagram of Figure 2.1 indicates, torque is converted by the motor transfer function into a rotational speed. This is a process that is equivalent to inertia filtering of the torque ripple that results from variable flux. For the purposes of this thesis, ripile is defined as the ratio of the amount of variation from the maximum to the maximum itself. Having made the appropriate substitutions in the model (see Appendix A for Revision Three), the program was run under no-load conditions. was chserved that there was, in fact, ripple in both the developed torque and motor speed and that some inertia filtering did occur. The next logical step was to load the motor and study the effect on torque and speed ripple. Table III indicates that under no-load conditions where speed was highest, the percentage of ripple in the output speed was at a minimum but that torque ripple was at a maximum.

It can be seen that as the load increased the amount of ripple in the developed torque decreased from about 84% to 8%. A closer examination of the torque behavior under varying loads revealed that, even as the motor torque necessarily grew with increasing loads, in every case there was

TABLE III
Torque and Speed Ripple Due to Sinusoidal Flux

MOTCR	MOTOR	% RIPPLE
IOAD (oz-in)	SPEED (RFM)	TORQUE SPEED
0.0	2557	33.9
32.0	2087	21.1
64.0	1617	12.2
96.0	1147	7.8

approximately a constant 8-9 oz-in variation. This variation can be attributed primarily to the flux variation in Ktpp and to a lesser degree to the variation in i(t) caused ty the variation in tack emf. Thus the magnitude of torque variation remained about the same for all loads though the percentage ripple appears to indicate otherwise. case of motor speed, the amount of ripple appeared to increase slightly with increasing load and thus decreasing speed. In this instance, the magnitude of variation did increase a small amount with decreasing speed - from 29 rpm at no-load to 52 rpm at 96 oz-ins - and suggests that speed ripple, at least, is somewhat dependent on load. due in part to the fact that as the motor slows, there is less rotor momentum and thus less inertia filtering. addition, at slower speeds there are less frictional effects which at higher speeds also tend to filter out ripple. general, it makes intuitive sense that at higher speeds frequency variations are harder to discern than at slower speeds.

D. HARMONIC FLUX

Having observed the effects of simple sinusoidal flux variation, the next step was to replace the single sinusoid with a flux composed of the sum of a sinusoid and its harmonics which is more representative of the air gap flux

patterns in an actual motor. The back emf of a typical commercial brushless dc motor is given in Figure 5.2 where the voltage between two terminals was measured with the motor rotating at 1200 rpm.

A harmonics analysis of this waveshape shows that the principle harmonic was the fifth and that the waveshape was closely described by the expression

3.0
$$\sin(\theta) + 0.59 \sin(5\theta)$$
 (eqn 5.14)

When substituted into equations 5.5 thru 5.7, and the different angular speed taken into account, the following relationships for flux result

$$\phi a = 3.0*\sin(2\theta + \pi/6) + 0.59*\sin(10\theta + 5\pi/6)$$
 (eqn 5.15)

$$\phi b = 3.0*\sin(20+9\pi/6) + 0.59*\sin(100+9\pi/6)$$
 (eqn 5.16)

$$\phi c = 3.0*\sin(2\theta + 5\pi/6) + 0.59*\sin(10\theta + \pi/6)$$
 (eqn 5.17)

The same procedures used in the simple sinusoid model were then applied to the harmonics case where the composite flux changes to $\phi v2$, a flux factor that varied to a slightly greater degree than did $\phi v1$. Equations 5.8 and 5.9 had to be modified as follows

$$Kbppp = Kb * (\phi v 2/KK3)$$
 (eqn 5.13)

$$Ktppp = Kt * (\phi V2/KK3)$$
 (eqn 5.19)

where KK3 approximately normalizes the flux and had the value KK3 = 5.38. As before, Kbppp and Ktppp varied with the flux and so the new equations for back emf and developed torque became

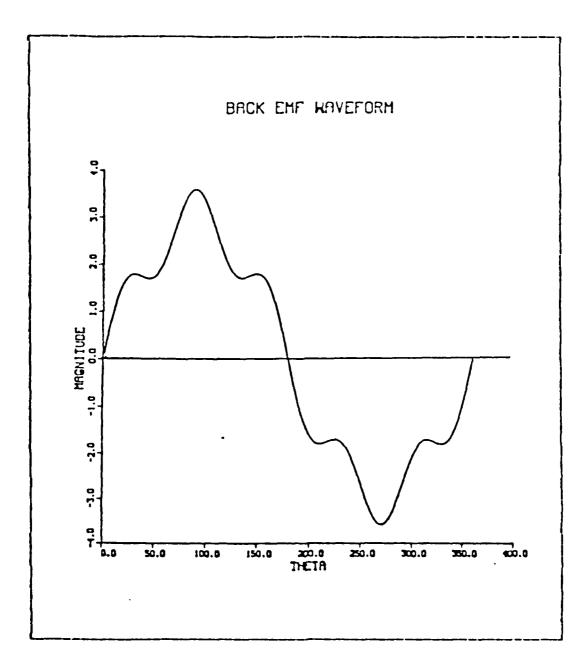


Figure 5.2 Back EMF Waveform Due to Harmonic Flux

$$eb(t, \phi v2) = Kb(\phi v2) * wm(t)$$
 (eqn 5.20)

$$eb(t, \phi v2) = Kbppp * wm(t)$$
 (eqn 5.21)

$$tm(t, \phi v2) = Kt(\phi v2) * i(t)$$
 (eqn 5.22)
 $tm(t, \phi v2) = Ktppp * i(t)$ (eqn 5.23)

The appropriate substitutions were once more made in the model (see Appendix A for Revision Four) and the model was run under various loads. As anticipated, the variable flux, ¢v2, produced ripple in the developed torque and motor speed. Table IV shows the resultant values for various loads.

TABLE IV
Torque and Speed Ripple Due to Harmonic Flux

MOTCR IOAD (oz-in)	MOTOR SPEED (rpm)	% RIPPLE SPEED
0.0	2557	94.6
32.0	2087	38.5
64.0	1617	22.5
96.0	1147	14.7

Comparing Tables III and IV, it can be seen that the values for torque and speed ripple are of similar magnitudes. The somewhat higher values for torque are due to the fact that in the ϕ v2 case there was a 16-21 0z-in variation as opposed to an average 8-9 cz-in variation in the ϕ v1 case. The speed ripple behavior was much like the ϕ v1 case with proportionate increases resulting from the increased torque ripple. Thus, for both torque and speed ripple due to ϕ v2, the model performed as expected; that is, slightly higher values of ripple occurred which were clearly attributable to the increased variability of the flux pattern of composite sinusoidal harmonics.

An important observation made during runs of the $\phi v1$ model came as a consequence of the motor speed variability,

particularly when the motor was unloaded. Since no-load speed was a function of Kb as given in equation 3.4. variable Khpp had the effect of increasing the upper-end speed beyond the 2557 rpm baseline given in Figure 3.1 This increased magnitude ccupled with the swing of speed values created an undesirable effect in the unloaded condition. In at certain points in time the additional speed produced back emf values that exceeded the input voltage which in turn resulted in negative current and developed torque values. Though these were transient negative values and had small effect on the steady state operation of the motor, still their presence was obviously unrealistic. solution to returning the motor to all positive values of current and torque was to increase the friction filtering effect by adjustment of the viscous friction coefficient, It will be recalled from Chapter Three that a value for the Bm of the commercial motor being modelled was not available and that for the basic model a value for Bm was derived from the curve fitting process. Thus, it seemed reasonable and permissable to repeat the procedure here.

Since it makes sense that increasing the friction within a system will slow the system down, in this case increasing the value of Bm did just that. By increasing Bm from 0.00015 to 0.045 oz-in/rad/s, the upper-end of the motor speed was reduced to a value that eliminated the undesirable negative values. This caused the no-load speed to be lowered slightly below the desired value of 2557 rpms. Recalling that no-lcad speed is fixed by the back emf constant, Kb, this parameter was adjusted from 0.1120 to 0.1089 vclt/rad/s to bring the back up to speed. Since this adjustment was less than 3% and the manufacturer specifications allowed a 10% +/- measurement error margin, adjustment to Kb was permissable. Of course, a proportionate adjustment to the torque constant, Kt, was also made

since Kt and Kb must maintain a constant relationship as explained in Chapter Three. The effect on motor speed ripple was neglible. Reviewing the \$\psi v\$2 model at this point revealed that similar behavior in the no-load state occurred but on a larger scale due to the greater variability of \$\psi v\$2. A similar set of adjustments was necessary to eliminate the negative current and torque values in the unloaded condition. It required an adjustment of Bm from 0.00015 to 0.045 oz-in/rad/s in order to ensure that all values were positive. In addition, the back emf constant, Kb, was adjusted to 0.1192 volt/rad/s and a proportionate adjustment to Kt was made. Again, all adjustments were within tolerances.

Of course, it was earlier established in Equation 3.7 that changes to Kb and Kt would change the slope of the speed-torque curve. Thus, in both cases the slope was returned to its original value by adjusting the motor resis-In the case of sinusoidal variable flux, R was tance, R. reduced from 2.74 to 2.70 ohms. For harmonic variable flux, R was increased from 2.74 to 3.0 ohms. Again, both adjustments were within the 10% +/- allowable deviation. also observed that since both Kb and Kt no longer remained constant but rather varied within small ranges, was not constant and varied accordingly in both the \$\psi v\$1 and ¢v2 models with the greatest change occurring when Kb and Kt simultaneously reached the minimum values of their respective ranges. Bearing in mind that these deviations in slope are fairly transient, their average values are most important and are the approximate values from which the final adjustments to R were based.

VI. SUMMARY

A. REMARKS AND CONCIUSIONS

CSMP language is a convenient tool for modeling a brushless DC motor. Positional sensor feedback and switching logic being functions of time, the action of electronic commutation can be nicely simulated in CSMP. The model can be fairly easily modified in order to study other motor configurations. For example, the three-phase star configuration used in this thesis could easily be changed to a grounded neutral, point D in figure 4.1, and the same types of analysis performed. In future studies where the complexity of the model is increased in order to include more design detail, the advantage of CSMP will become even more apparent.

In the course of simulating the effects of different air gap flux variations, it appeared necessary to make minor parameter adjustments in order to offset the feedback effects of torque and speed ripple. Within the scope of this thesis, adjusting the steady state performance was reasonable and justifiable. In a larger context, however, various control systems, particularly torque generation are specifically designed to reduce torque controllers. ripple and its effects. So, in at least one sense, these adjustments were somewhat artificial. To the extent that parameter adjustment is necessary and desirable, it must be understood that this is not a simple procedure since adjustment of one parameter almost always requires adjustment of one or more others. These adjustments in turn necessitate readjustment of the criginal parameter. The process is thus an iterative one. This model permits this kind of parameter adjustment, but it is a long and tedious process that would be better performed by some sort of optimization algorithm.

E. RECCHMENDATIONS FOR FUTURE STUDY

The simulation of ripple effects is particulary relevant to the control engineer who must design the control circuitry and power conditioner to meet specific performance requirements. This points to several areas for future study. One area is the control of torque generation and the several principle configurations used in brushless DC motors. Two commonly used methods are the sinusoidal torque generation principle and the trapezoidal torque scheme. In addition to torque generation control, second area for investigation is power control of brushless DC motors. In this thesis, output power was unregulated and was studied simply as a function of a constant input however, power is produced and voltage. In practice, controlled by varying the supply voltage. Tarious schemes exist for power control; amoung these are linear transistor control and pulse-width or pulse-frequency [Ref. 5].

A very important area that this thesis did not address was that of the switching transients that occur as a consequence of commutation. During each switching interval, each winding is an inductor that stores energy which causes a significant amount of current to flow at the instant of switching. This can have serious effects, especially if breakdown conditions exist amoung the commutation transistors and other control circuit elements. In this same area, switching transients also occur within the control circuitry itself. Thus, a thorough investigation and understanding of the behavior of power transistors, diodes and other control elements used with electronic commutation needs to be done.

APPENDIX A LISTING OF MODEL PROGRAMS

A. BASIC PROTOTYPE MODEL

The program in this section is a basic program that simulates the action of a standard brush-type DC motor. It is the prototype from which the brushless DC motor model and its several revisions are derived.

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BRUSHLESS
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11
                                                                                                                                                                                                                                                                                                                                                                                                                                    * KT -- TCRQLE CONSTANT (CZ-IN/AMP)

* KB -- EACK ENF CONSTANT (VCLT/RAD/S)

* RA -- RESISTANCE OF THE MOTCK (OHM)

* BM -- VISCCLS FRICTICN COEFFICIENT OF THE MCTOR

* BL -- VISCCLS FRICTION COEFFICIENT OF THE LCAD

* BL -- VISCCLS FRICTION COEFFICIENT OF THE LCAD

* BL -- VISCCLS FRICTION COEFFICIENT OF THE LCAD

* BL -- VISCCLS FRICTION COEFFICIENT OF THE LCAD

* DM -- TOTAL VISCOUS FRICTION COEFFICIENT OF THE LCAD

* JL -- INERTIA OF THE ELECTRICAL TIME CONSTANT OF THE

* AZ = JP/RM -- THE ELECTRICAL TIME CONSTANT OF THE
                                                                                                                                                                                                                                                                               INITIAL
CONSTANT KT = 15.5, BM = 0.00015, BL = 0.0, JL =0.0,
N = 1.0, JM = 0.0016, KB = .112, PI = 3.14155265
PARAMETER LA = .0016, RA = 2.74, ...
                                                                                                                                                               THIS IS THE BASIC PROGRAM THAT STANDARD BRUSH-TYPE DC MCTOR. FROTCTYPE FROM WHICH ALL LATER ICC MOTOR MODELS ARE DERIVED.
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THIS PRCCECURE RESETS THE VARIABLE THRST TC O AFTER EVERY 360 CEGREES OF MECHANICAL RCTATION. THIS IS FUNDAMENTAL TO THE SIMULATION OF ALL SWITCHING ARE POSITION SENSING ACTION.
                                                                                                                                                                                                        THIS PROCECURE PROVIDES A SIMPLE MECHANISM FOR FEVERSING THE MOTOR'S CIRECTION
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                C POTOR SYSTEM
= .04, GUICEL = 0.0004, PRCEL
M, TM, MM, WMRPM, PWR
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   PROCEDURE THRSI=RESET(JFAC, TLUEG)
15 = JFAC * 360.0
1hRSI = 1LCEG - TS
1F(THRSI LI.340.0) GC TC 40
2FAC = JFIC * 1.0
40 CONTINUE
ENDPRECEDURE
                                                                                                                                                                                                                                                         PROCEDURE TN1=FACEMD(VIN, TM, TL)
IF (VIN, LT, C, 0) GO TG 10
IN1 = TM - TL
GG TC 15
10 TN1 = TM + TL
15 CONT INUE
ENDPFECECURE
                                                                                                                                     | * KE
| NIGRL (C. 0 WM)
| * IP * .0073615
VI = 30.0 * SIEF(0.0)

VIN = VI - VEMF

VIN = VIN | * (1.0/RA)

IM = REALPL(C.0.AI,VIN2)

IN = IN * KI

IN = TNI * (1.0/BM)

WMM = REALPL(C.0.AZ,IN2)

WMM FPM = WM * (30.0/PI)

VEMF = WM * KE

IHETA = INICAL(C.0,WM)

PWR = MM * IM * C.0,WM)
STEF(0.0)
VEMF
(1.0/RA)
(C.0,AI,VINZ)
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```

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B. REVISION ONE

This is the first revision of the basic prototype. This model simulates the switching logic of a brushless DC motor which is based on feedback from Hall effect sensors. In addition, the windings are treated independently and their contributions to developed torque are superposed. Since the superposition results in twice as much current flow as a single lumped coil, the total current is halved to retain the same overall motor behavior as in the prototype.

THIS IS THE FIRST REVISION OF THE BASIC PRUTOTYPE. PODEL SINULATES THE SWITCHING LOGIC OF A BRUSHLESS MUTOR WHICH IS BASED ON FEEDEACK FROM HALL EFFECT SENSORS AS DESCRIBED IN CHAPTER FOUR. IN A DLITITION, THE WINDINGS ARE TREATED INCEPENCENTLY AND THEIR SUPERPOSABLE. SINCE THE SUPERPOSITION RESULTS IN THE TOTAL CURRENT FLOW AS A SINGLE LUMPEL CUIL, THE TOTAL CURRENT IS HALVED IN CROSE TO KEEP THE CVERALL BEHAVIOR THE SAME AS IN VERSION CNE. 61, * THUM AS-L AB6 * , CLA SS=C \$911 \$72p 「ころろく GVM1.2 // THCCSM P2 JCE C. /*#AIN CRG=NFGV. // EYEC CSMPX! //X.COMPRINT CC //X.cYSPFINT CC //X.cYSPFINT CC VERSION INC

5205 = (0.0, 32., 64., 96., 128) C. 00015 BL PI = 3.14155 = 2.740, TL KB = 3.112; F = .0015; RA = INITIAL CONSTANT KT JM = 0.001

KT -- TCRULE CONSTANT (CZ-IN/APP)

RA -- RESISTANCE OF THE MOTOR (GHM)

BM -- VISCUS FRICTION COEFFICIENT GF THE MCTOR (OZ-IN/RAD/S

BL -- VISCUS FRICTION COEFFICIENT GF THE LCAD

BLF -- VISCUS FRICTION COEFFICIENT GF THE LCAD

BL -- TCTAL VISCUS FRICTION CEFFICIENT GF THE MOTOR SYSTEM

-- INERTIA GF THE MCTOR (UZ-IN/S-S)

JLF -- INERTIA GF THE LCAD

JLF -- INERTIA GF THE LCAD

JLF -- TOTAL INERTIA CF THE LCAD

JLF -- TOTAL INERTIA CF THE LCAD

JLF -- TOTAL INERTIA CF THE COORTANT OF THE MOTOR

AI = LA/RA -- THE MCTOR TIME CONSTANT OF THE MOTOR

AZ = J/P -- THE MCCHANICAL TIME CONSTANT OF THE MOTOR

GEARS

NOSGFT BLF = EL/(N**2) JLF = JL/(N**2) J = JM + JLF B = EM + BLF Al = LA / FL A2 = J / B

THRST = 0.C JF &C = 0.0 DYNAPIC VIF = 30.0 * STEF(0.0) VIR = C.0 VIN = VIN + VIB VIN = VIN + VERF VIN = VIN + VERF VIN = VIN + VIB IME = REALFI(0.03A1.VA) IME = REALFI(0.03A1.VB) IMI = IMA + IME + IMC //2 IMI = IN + VIN + VINC //2 IMI = IN + VIN + VINC //2 IMI = IN + VINC THIS PROCECURE PROVIDES A SIMPLE MECHANISM FOR REVERSING THE MOTUR'S DIFFECTION.

PROCEDURE IN 1 = F & D EWD (VIN, TM, TL)

IF (VIN, LT, C, C) GO TC 10

IN 1 = TM - 16

C TC 15

10 IN 1 = TM + TL

15 CCNT INUE

END FCCECURE

EVERY 360 DEGREES SIMULATION OF ALL * THIS PRCCECLRE RESETS THE VARIABLE THRST TC O AFTER * OF MEC.FANICAL RCTATIGN. THIS IS FUNDAMENTAL TO THE * SWITCHING ARE PESITIEN SENSING ACTION.

PROCEDURE THEST=RESET(JFAC, TEDEG)
15 = JFAC + 340.0
1HRST = TELEG - TS
1F(TERST.LT.360.0) GC TC 40

```
TERMINAL

TITLE EASIC CC MOTGR SYSTEM

TIMER FINITAL = .040, DUTDEL = .0004, PRCEL = .0004

PRINT TERSTRUEMFILMA, IME, IMC, IMT, TM, MM, HMRPW

* PRINT SEI, SEZ, SE3, Shi, Sw3, Sw4, Sh5, Sh6

LABEL MCTGR SPEED DUE TC STEP INPUT

* PAGE XYPLCT

ENDJCB STCP

ENDJCB
```

END P CELTIN BENDER PROPERTY OF THE PROPERTY O

C. REVISION TWO

In the following program, the motor's air gap flux is treated as the average value of the sum of the flux acting in two windings as explained in Chapter Five. For convenience of study, both windings are again lumped into a single winding as in the prototype. This facilitates a closer focusing on the effects of variable flux on motor behavior.

1S TO TREAT THE AIR C. CLUX ACTING IN TWO CONVENIENCE OF STUDY, I RSION CNE IN ORDER TO TABLE FLUX CN MOTOR * KT -- TERCLE CENSTANT (CZ-IN/AMP)

* KB -- BACK EMF CONSTANT (VELT/RAD/S)

* RA -- FESISTANCE OF THE MCTER (GHM)

* BM -- VISCELS FRICTION COEFFICIENT OF THE CAD

* BLF -- VISCELS FRICTION COEFFICIENT OF THE CAD

* BLF -- VISCELS FRICTION COEFFICIENT OF THE MOTER

* B -- TCTAL VISCEUS FRICTION COEFFICIENT OF THE MOTER

* JL -- INERTIA OF THE LLAD THRU REDUCTION GEARS

* JLF -- INERTIA OF THE LLAD THRU REDUCTION GEARS

* JLF -- TOTAL INERTIA OF THE MCTOR SYSTEM

* JLF -- TCTAL INERTIA OF THE MCTOR SYSTEM

* JLF -- TOTAL INERTIA OF THE MCTOR SYSTEM

* JLF -- TOTAL INERTIA OF THE MCTOR SYSTEM

* JLF -- TOTAL INERTIA OF THE MCTOR SYSTEM

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* TL -- TOTAL INERTIA OF THE MCTOR SYSTEM

* TL -- TOTAL INERTIA OF THE MOTOR

* THE MCRMALIZED PRODUCT OF KT AND PHIAVG

* KTF -- THE MCRMALIZED PRODUCT OF KT AND PHIAVG

* KTF -- THE MCRMALIZED PRODUCT OF KT AND PHIAVG GEARS 18 X: 01 C.O, N = 1.631, INITIAL CONSTANT KT = 15.5, BM = 0.00015, BL = 0.C, JL = JM = 0.CO1, kB = 0.112, PI = 3.14159265, PHIAVG PARAMETEF LA = 0C16, RA = 2.740, 128.0) LEKSION THREE - THIS PURPOSE OF THIS PROGRAM!
LLX AS THE AVERAGE VALUE OF THE SUM OF THE FI
IINDINGS AS EXPLAINED IN CHAPTER FIVE. FOR CO
IINDINGS ARE LUMPED INTO ONE WINDING AS IN VER
ACILITATE CLOSE STUCY OF THE EFFECTS OF VARIA (2672,011 GVM1.2672P CLMMY DLMMY ** THECSMP2 CCE | ** THE CC CMB | ** CCMB | CC CMB | CC C >43340

59

A2 = J / B THFST = 0.C JF &C = 0.0 KTF = KT * FFIANG * (1/KK1) KBF = KE * FFIANG * (1/KK1)

DYNAFIC VIF = 30.0 * STEF(0.0) VIR = 0.0 VIN = VIN - VERF VIN = VIN - VERF VIN = VIN + (1.0 /RA) IMF = REALF(0.0 A1, VIN 2) IMF = RALF(0.0 A2, IN 2) WM = REALF(0.0 A2, IN 2) WM FFM = WM + (30.7 FI) WM FFM = WFFFM/N WM FFM = WFFFM/N THETA = INTEFL(0.0, WM) THECO = THETA + (180.0 /FI) THECN = THRST THECN = THRST VEFF = KBP * WM SIMPLE MECHANISM FOR REVERSING THE MCTUR'S * THIS PRCCECLEE FROVICES A * DIFECTION.

PROCEDURE IN 1=FADBWD (VIN, TM, TL)

IN 1 = TF - TL

CO TC 15

10 1N1 = TM + TL

15 CONTINUE
ENDPRECECURE

THIS PROCECURE RESETS THE VARIABLE THRST TO AFTER EVERY 340 DEGREES OF MECHINICAL ROTATION. OF ALL SWITCHING AND POSITION SENSING ACTION.

PRUCEDURE THRS1=RESET(JFAC, THDEG)
1S = JFAC * 360.0
1HRS1 = 1FCEG - TS
1F(TFRST.LI.360.) GO 10 40

E ACTICA THE AND THE LUGIC mm_F RESTO HE VARIABLES SEL TI CF TFIS SIMULATE CFF IN THE FORMER EC OR SMITCHED OFF GF WHICH THE SWITCH 2,5W3,5W4,5h5,8W6=VOLT(ThCON 01235 SUNDING フリンフリン C SET THE LS 1 CR O D ON CR O ENERGIZED THRCUGH 200000 000000 LEVEL LEVEL URNE ING I ABL -----SW6 TC LCGIC L SSW6 TC LCGIC L NSORS BEING TU RANSI SIORS BEI ON IS THE VARI 3 SW 1 SW 2 45 AAAAAA VANNAN ODDOOO 180, SE3 AND SMITHAL SWOOD OF HALL EFFECT SENSONS SIMULATES FOMER TRAIL LATTER CASE THOON | SE2,SE3 200000 200000 200000 ŧ 1+CCN 40 CONTINUE END FECELLE

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Shi = 0:

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Sha = 0:

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D. REVISION THREE

The purpose of this version of the program is treat the air car flux as varying in simple sinusoidal fashion. The total system flux is treated as the algebraic sum of the flux developed in the two active windings.

```
FLUX
                                                                                                                                                                                                                                            VERSION FOLF -- THE PURPOSE OF THIS VERSION OF THE PRUGRAM IS TO TREAT THE AIR GAP FLUX AS VARYING IN SINUSCIDAL FASHION. THE TOTA SYSTEM FLUX IS TREATED AS THE ALGEBRAIC SUP OF THE FLUX DEVELOPEITHE TWO ACTIVE WINDINGS AS EXPLAINED IN CHAFTER FIVE.
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                     GEARS
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                         VAR I ABL E
VAR I ABL E
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                     * KI -- TCRCLE CONSTANT (VCLT/RAD/S)

* KB -- EACK EFF CONSTANT (VCLT/RAD/S)

* RA -- RESISTANCE OF THE MOTCR (GHM)

* BM -- VISCELS FRICTION COEFFICIENT OF THE LCAD

* BL -- VISCELS FRICTION COEFFICIENT OF LCAD

* BLF -- VISCELS FRICTION COEFFICIENT OF LCAD

* JN -- TCTAL VISCEUS FRICTION CEFFICIENT OF LCAD

* JL -- INERTIA OF THE MCTOR SYSTEM

* JL -- INERTIA OF THE LCAD

* JL -- INERTIA OF THE MCTOR SYSTEM

* JL -- INERTIA OF THE MOTOR

* JL -- INERTIA OF THE MOTOR

* AZ = J/C

* JF -- THE NORMALIZED PRODUCT OF KB ANC THE COMPOSITE VARIABL

* KBFP -- THE NORMALIZED PRODUCT OF KB ANC THE COMPOSITE VARIABL

* KTFP -- THE NORMALIZED PRODUCT OF KB ANC THE COMPOSITE VARIABL
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                          (02-IN/RAD/SI
                                                                                                                                                                                                                                                                                                                                                                                                                                                                      Z
                                                                                                                                                                                                                                                                                                                                                                                                                              INITIAL
CONSTANT KT = 15.43, BM = 0.022, BL = 0.0, JL = 0.0,
JM = 0.001, KE = 0.1089, PI = 3.14155265
PARAMETER LA = .0C16, RA = 2.700,
KK2= 1.6551, TL = (0.0, 32.0, 64.0, 56.0, 128.0)
//THCCSMF2 JCE (2472,0114), TFUMAS-LAB6", CLASS=C/*MAIN CFG=NFGVM1.2672P
// EXEC CSMPX\
//X.COMPRINT CC CLMMY
//X.SYSPRINT CC CLMMY
//X.SYSPRINT CC CLMMY
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                           38.9
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DYNAPIC VIE = 3C.0 * STEF(0.0) VIN = VIF + VIB VIN = VIF + VIB VIN = VIN + VIB VIN = VIN + VIB THE = REALF (0.0 A 1. VIN 2) THE = REALF (0.0 A 1. VIN 2) THE = THE P (0.0 A 2. THE TA + (1.0 PIN 6) BEFFC = SIN (2. * THE TA + (1.0 PIN 6) BEFFC = SIN (2. * THE TA + (1.0 PIN 6) BEFFC = SIN (2. * THE TA + (1.0 PIN 6) THE TA = THE TA + (1.0 PIN 6) THE C = THE TA + (1.0 PIN 6) THE TA + (1.0 PIN 6) THE C = THE TA + (1.0 PIN 6) THE TA + (1.0 PIN

1

THIS PRCCECURE PROVICES A SIMPLE MECHANISM FOR REVERSING THE MGTOR *S DIRECTION.

PROC ELURE IN 1= FMCEWD (VIN, TM, TL)

IF (VIN, LT, 0,0) GO TG 10

IN 1 = TM - TL

CC TC 15

10 TN 1 = TM + TL

IS CONTINUE

ENDPFOCE CLRE

* THIS PRCCECURE RESETS THE VARIABLE THRST TC O AFTER EVERY 340 DEGREES * OF MECHANICAL RCTATION THIS IS FUNDAMENTAL TO THE SIMULATION OF ALL * SWITCHING AND POSITION SENSING ACTION.

PROCEDURE THRST=RESETLJFAC, THDELJ 15 = JFAC * 360.0 THRST = TFLEG - TS

THE ACTION CASE AND IN THE 0. <u>10</u>61C THE PURFOSE OF THIS PROCEDURE IS TO SET THE VARIABLES SELTHRISES AND SWITHRUS WE TO LOGIC LEVELS IOR OF THIS SIMULATES THOUGH HALL EFFECT SENSORS BEING TURNED ON OR OF IN THE FORMER SIMULATES FOWER TRANSISTORS WEING ENERGIZED OR SWITCHED OF LATTER CASE. THOOM IS THE VARIABLE THROUGH WHICH THE SWITCHIIS IMPLEMENTED. 01 09 16 TERST - LT - 340.)
FAC = JFAC + 1.0
CONTINUE
ENDP FGCECLRE ***

V ARIABLES AS THE SUM OF MCTUR'S EASILY THAT VAL VARIA FERM AND EFFECT AS NE THE ALGEBRAIC SI IN CUMPUTING THE MI UNCTION COLLD AS E-PROCEDURE EXCEPT TO DATE THE ACCITIONAL G PRUCEDURE HAS THE SAME FCR FLLX CCMPCNENTS FOR USE IN EACK EMF (BEMFT). THIS FUNC ICULATED IN THE PREVIOUS PRO SUFFICIENT GCM TO ACCOMUDAT THE FOLLOWING PREVIOUS. IT THE VARIABLE APPRCXIMATE E HAVE BEEN CAL THERE NAS INS

E. REVISION FOUR

This version of the model treats the flux as varying according to the sum of a sinusoid of fundamental frequency and its fifth harmonic as explained in Chapter Five. The total flux is again approximated as the algebraic sum of the flux developed in two windings at a time.

```
HAPTER
THE FLUX
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                          RIABLE
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                       * KT -- 1CRCLE CCNSTANT (VCLT/RAD/S)

* KB -- BACK EMF CON STANT (VCLT/RAD/S)

* RA -- RESISTANCE OF THE MCTCR (OHM)

* BM -- VISCCLS FRICTION COEFFICIENT OF THE MCTOR (UZ-IN/RAD/S)

* BLF -- VISCCLS FRICTION COEFFICIENT OF THE MCTOR (UZ-IN/RAD/S)

* BLF -- VISCCLS FRICTION COEFFICIENT OF THE MCTOR (UZ-IN/RAD/S)

* BLF -- VISCCLS FRICTION COEFFICIENT OF THE MCTOR (UZ-IN/RAD/S)

* BLF -- VISCCLS FRICTION COEFFICIENT OF THE MCTOR (UZ-IN/RAD/S)

* BLF -- VISCCLS FRICTION COEFFICIENT OF THE MCTOR SYSTEM

* JLF -- INERTIA OF THE MCTOR (UZ-IN/S-S)

* JLF -- INERTIA OF THE MCTOR (UZ-IN/S-S)

* JLF -- TOTAL VISCCUS FRICTION GEARS

* JLF -- TOTAL VISCOUS FRICTION GEARS

* JLF -- TOTAL VISCOUS FRICTION GEARS

* JLF -- TOTAL VISCOUS FRICTION GEARS

* JLF -- TO
                                                                                                                                                                                                                                                                                                                                                                                                                                                                    VERSION FIVE -- THE PURFGSE OF THIS VERSION OF THE PROGRAM IS TREAT THE FILX AS VARYING ACCORDING TO THE SUM CF A SINUSUID FUNCAMENTAL FRECUENCY AND ITS FIFTH HARMONIC AS EXPLAINED IN FIVE. THE TCTAL FLUX IS APPROXIMATED AS THE ALCEBRAIC SUM CFDEVELUFED IN THE WINDINGS AT A TIME.
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                           INITIAL CONSTANT KT = 16.82, BL = 0.0, JL = 0.0, h = 1.0, ...
JM = 0.001, kE = 0.11875, PI = 3.14159265
PARAMETER LA = 0016, RA = 2.624, 1 = (0.0,32.,64.,96.
//THCCSMP2 JCE (2672,0116), ThOMAS-LAB6"; /**MAIN CRG=NFGVM1.2672P // EXEC (SMP X) // X. CCMPRINT LC CLMMY // X. SYSPRINT CC CLMMY // X. SYSPRINT CC CLMMY // X. SYSPRINT CC CLMMY // X. SYSIN DC *
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                       NUSCRI
BLF = EL/(N**2)
JLF = JL/(N**2)
J = JM + JLF
B = BM + BLF
Al = LA / FF
```

59#SIN(10 59#SIN(10 59#SIN(10 (2.*THET A+(1*PI/6))+ (2.*THET A+(9*PI/6))+ (C.O.*HET A+(5*PI/6))+ (C.O.*M) - VIB - VEMF | * (1.0/RA) F(0.0.4)A1 (C-0,A2,TN2) STEF(0.0) \\
\(\lambda \) \\
\(A2 = J THFST : JFAC = :

FETA+(5*P1/6))) FETA+(9*P1/6))) FETA+(1*P1/6)))

THIS PROCECURE FROVIDES A SIMPLE MECHANISM FOR REVERSING THE MCTOR'S DIRECTION.

PROCECURE TN = F & DEWD (VIN, TM, TL)

IF (VIN, LT, C, C) GO TO 1C

IN = TM - TL

GO TC 15

10 IN 1 = TM + TL

15 CONTINUE

ENDPRECEEURE

EVERY 300 CECREES SIMULATION OF ALL THIS PROCECURE RESETS THE VARIABLE THRST TO DAFTER OF MECHANICAL ROTATION. THIS IS FUNDAMENTAL TO THE SWITCHING APE POSITION SENSING ACTION.

PROCECURE THEST=RESETIJFAC, THEEG)

* .0 0706155 * NM BENT * (1/KK2) BENT * (1/KK2)

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THE ACTION
CASE AND
IN THE
                                                          THE PURFOSE OF THIS PROCEDURE IS TO SET THE VARIABLES SET THRIES AND SWITCH SWOLD TO LEGIC LEVELS I OR 0. THIS SIMULATES THE HALL EFFECT SENSORS BEING TURNED ON OR OFF IN THE FORMER CAST PULATES FOWER TRANSISTORS BEING ENERGIZED OR SWITCHED OFF IN ATTER CASE. THOUN IS THE VARIABLE THROUGH WHICH THE SWITCHINS IMPLEMENTED.
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1.00
1.00
1.00
1.00
TS = JFAC +
THFST = THC
IF (THR ST - L1
JFAC = JF
40 CCN IN C
END FCCECRE
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EFFECT AS THE LGEBRAIC SUM GF TING THE MCTOR'S OULD AS EASILY E EXCEPT THAT IT THE ALC CONFUT HAS THE SAME FC RE TO CETERMINE ENTS FCR USE IN MFI). THIS FUNC THE PREVIOUS PR NECT IN THE PROPERTY OF THE PR ING PROCEDURE 11 1S USED F LE FLUX CCMPC E EACK EMF (B THE FULLOWING PREVIOUS 11 THE VARIABLE APPRCXIMATE E HAVE BEEN CFL

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# CCLLD NCT ACCEMENTE THE ACCITIONAL VARIABLES.

PROCEDURE

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66 CHICLE

66 CHICLE

16 TILL

60 TICL

66 CHICLE

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17 TILL

18 TIL
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APPENDIX B SAMPLE OUTPUT

The following is a sample of the type of information the model can provide. In this particular case, the fourth revision of the basic prototype, the harmonic flux case, was used. For completeness, the motor was run at loads from 0.0 up through 96.0 oz-in, which was the range of loads with which all analysis was performed. In addition to such motor variables as speed, current and power, a sampling of the feedback from Hall effect sensors as well as the switching action of the power transistors is included.

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LIST OF REFERENCES

- 1. Fiscal Year 1983 NAVAIR Strike Warfare Technology Plan, Advanced Missile Control Devices, by R. F. Dettling, September 1982.
- Speckhart, F. H. and Green, A. I., A Guide to Using CSMP The Continuous System Modeling Program, Prentice-Hall, Inc., 1976.
- 3. Kuo, B. C., <u>Automatic Control Systems</u>, 4th ed., Frentice-Hall, Inc., 1982.
- 4. Fitzgerald, A. E. and Kingsley, Jr., Charles, <u>Flectric</u>
 <u>Machinery</u>, McGraw-Hill Bock Company, Inc., 1952.
- 5. <u>D.C. Motors, Speed Controls, Servo Systems, An Engineering Handbook, 4th ed., Electrocraft Corporation, 1578.</u>

BIBLICGRAPHY

Demerdash, N.A., Miller, R.H., Nehl, T.W., "Comparison Between Features and Ferformance Characteristics of Fifteen HP Samarium Cobalt and Ferrite Based Brushless DC Mctcrs Operated by the Same Power Conditioner," IEEE Transactions on Power System Apparatus and Systems, v. PAS-102, January 1983.

Demerdash, N.A. and Nehl, T.W., <u>Dynamic Modeling of Brushless</u>

<u>DC Motor-Power Conditioner Unit for Electromechanical Actuator Application</u>, paper presented at IFEF Power Electronics Specialists Conference, San Fiego, California, 9 June 1979.

Miller, T.J., "Rare Farth Magnets Contribute to Small Size, High Torque of New Motor Designs," <u>Control Engineering</u>, May, 1983.

Murugesan, S., "An Overview of Electric Motors for Space Applications," IEEE Transactions on Industrial Electronics and Control Instrumentation, v. IECI-28, November, 1981.

National Aeronautics and Space Administration Report cr-160349, Numerical Simulation of Dynamics of Brushless DC Motors for Aerospace and Other Applications, by N.M.O. Demerdash and T.W. Nehl, 15 November 1979.

National Aeronautics and Space Administration Report tm-80445, Analytical Modeling of the Dynamics of Erushless DC Motors for Aerospace Applications, A Conceptual Framework, by N.A.O. Demerdash, F.E. Eastman, and R.G. Chilton, 18 August, 1976.

Oppenheimer, M., "In IC Form, Hall-Effect Devices Can Take On Many New Applications," <u>Electronics</u>, August 2, 1971.

Society of Automotive Engineers, Inc. Technical Paper Series, #780581, Flectromechanical Actuator Technology Program, by J.T. Edge, April 1978.

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